



European Low Emission Combustion Technology

CA ELECT - AE

An overview

Duration: 1.1.2005 – 31.12.2008

Budget: 1,49 MEUR

Coordinator: Rolls-Royce Deutschland

START Brokerage Event – Riga - Latvia – 20 April 2005

Presented by: R. v.d.Bank / RRD

ALSTOM

Avio
propulsione aerospaziale

MTU
Aero Engines

Rolls-Royce

Sneema Moteurs
groupe sneema

Turbomeca
sneema group

DLR

ONERA



European Low Emission Combustion Technology

What are the objectives of the project ?

Implementation of the ACARE goals / Vision 2020 on Emissions

- **Strategy on How-To-Do combustor technology development**
- Integration & strengthening of the European Research Area
- Enhance exploitation in Europe
- Dissemination of European Research results and exchange of information in Europe
- Active search and identification of appropriate SMEs and of capable research partners from the new EU member states



European Low Emission Combustion Technology

European Aeronautics 2001 (SRA1) / The Challenge of the Environment (perspective of 20 years)

Objectives of ACARE (Advisory Council on Aeronautics Research in Europe)'s Vision 2020 / Strategic Research Agenda 2

1. Reduce NO_x by 80%

Interpretation: 80% LTO cycle relative to CAEP/2 & maintain current level of CO, soot, UHC

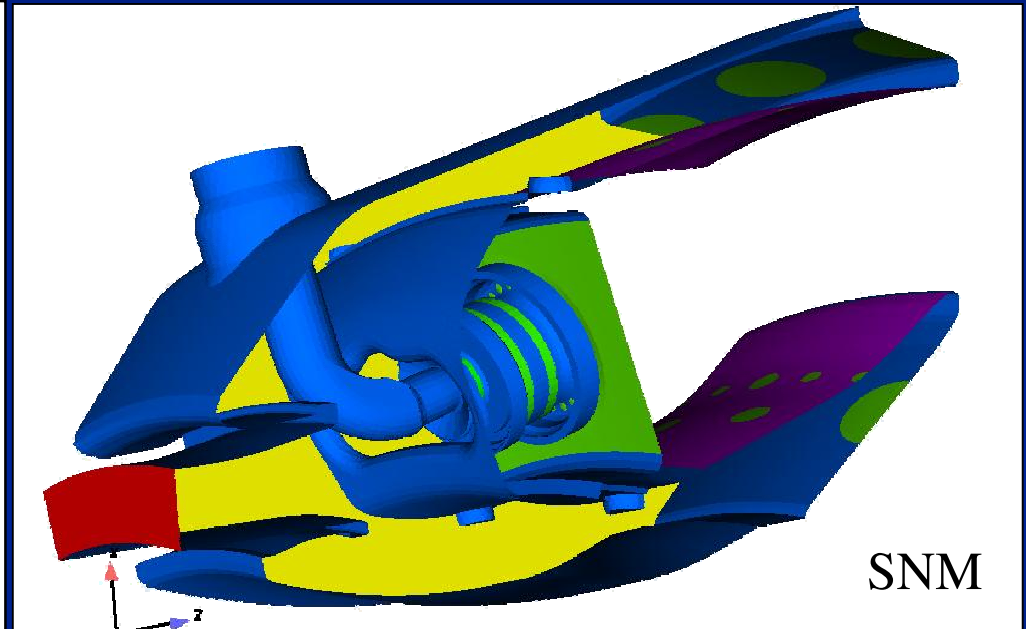
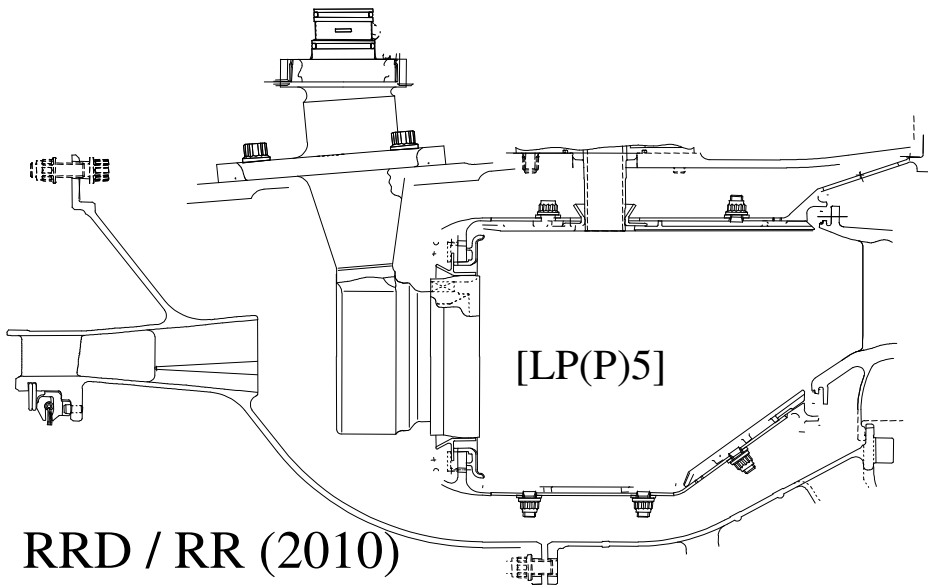
2. Reduce fuel consumption and CO₂ emissions by 50%
(P30↑ T30↑ ⇒ NO_x↑)
(BPR↑ OAFR↓ ⇒ NO_x↑)

Contribution split: 20-25 % airframe / 15-20 % engine / 5-10 % operations & air traffic (ATM)



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Low NO_x Single Annular Combustor Concepts in Europe Medium Term Perspectives



- ➔ Efforts have to be increased to cope with the challenge
- ➔ Economic growth requires technological innovation

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Interdependencies

Architecture:

RQL
DAC
ASAC
Internally Staged

Injectors:

Air Blast
LP(P) & LDI

Fuel Injection

Spray Break-Up
Spray Vaporisation
Droplet Diameters

Pressure Loss
Air Feeding
Pre-Diffuser

Fuel Coking
Fuel Supply
Schedule
Fuel Types
Aromatics

Operability:

Cold start, Ignition
Altitude Relight
Hail & Rain
Slam Deceleration
Weak Extinction

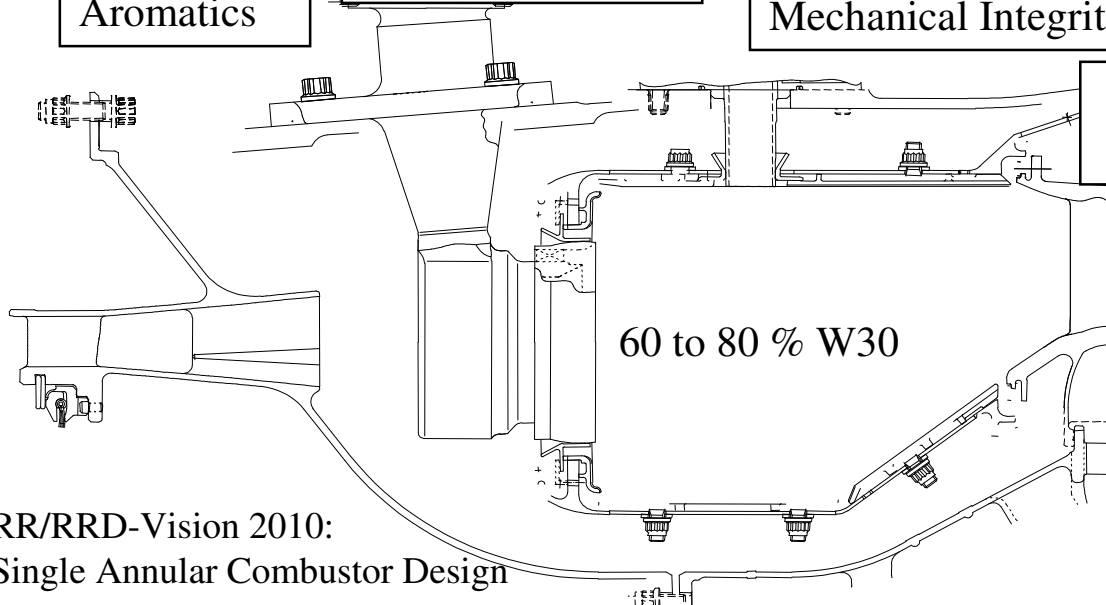
Costs
Competitiveness

Weight
Length
Part Count
Complexity

Burn-out
Combustion Efficiency

Thermo-Acoustic Instabilities
Combustion Driven Pressure Oscillations
Mechanical Integrity / Lifting

Temperature, Emissions
Chemistry, Exit Profiles



Emissions

NOx
Soot
CO
UHC

Liner Cooling / Lifting
Surface / Volume Ratio

- Low NOx design is highly complex and requires substantial amount of efforts and resources to address all key elements
- Extensive leading edge research is required





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Conclusions

Ultra-low NOx aero-combustion technology is highly challenging

Cost reduction & competitiveness / pre-requisite / successful implementation

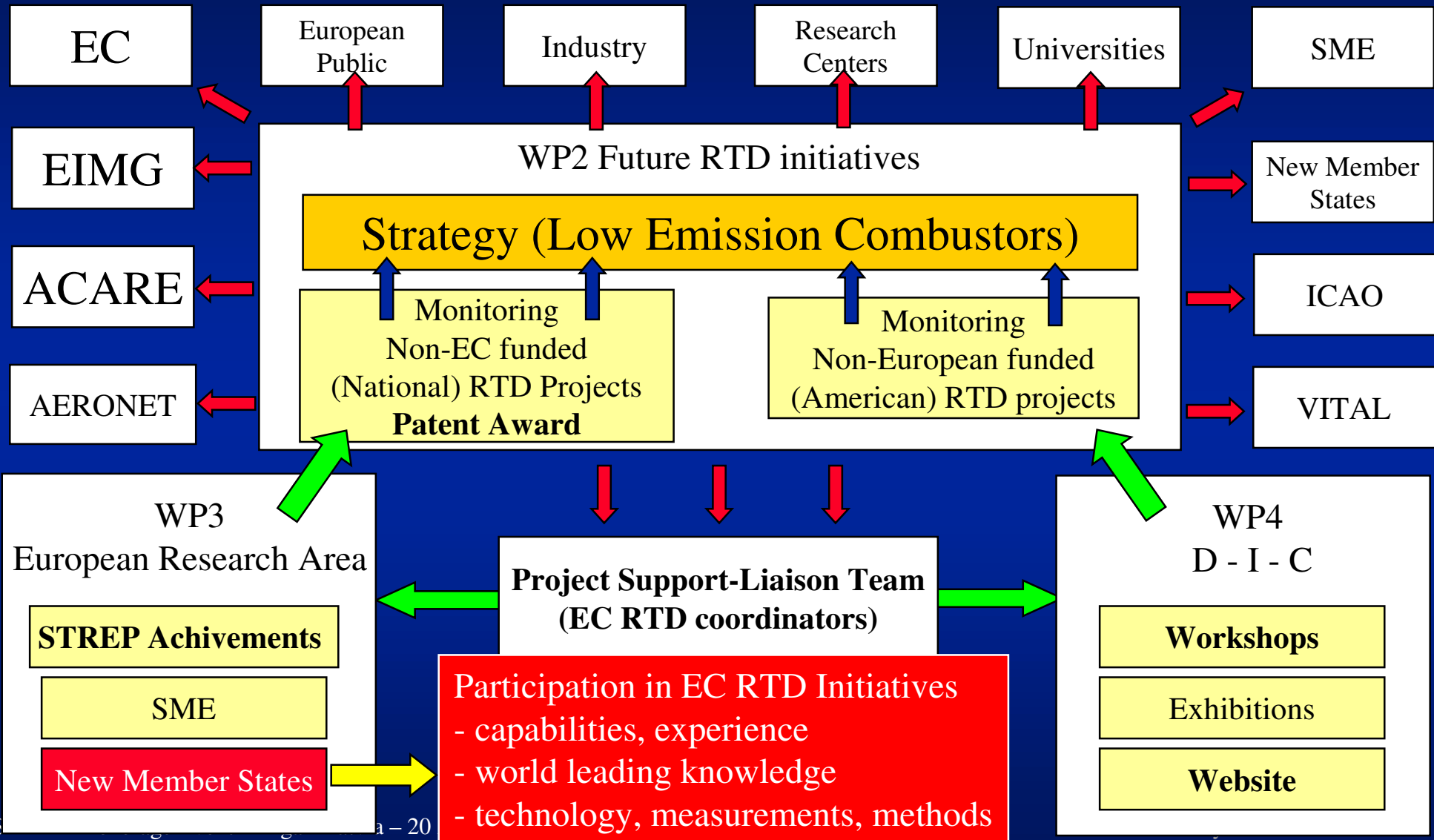
Focus: • Technology • Methods & Tools • Design Methodologies

Major technological issues:

- Emissions (Nox / Soot)
- Operability
- Thermo-acoustic high-amplitude pressure oscillations
- Cooling / Lifting



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Method Development

STREP TimeCop-AE
Towards Innovative Methods
for Combustion Prediction in
Aero-Engines

Turbomeca, Pau
(Lorenzo Hernandez)

AVIO, MTU, RR, RRD, SNM
NMS: Czenstochowa (PL)

LES / URANS development
2-phase flow with combustion
Numerical and experimental
validation of models

Budget: ~ 8,5 MEUR
EC funding: ~ 70 %
2006 – 2010 (48 months)

RTD Initiatives

FP6 C3

Fundamental Knowledge Generation

STREP TECC
Technology Enhancement for
Clean Combustion

Snecma Moteurs, Villaroche
(Michel Cazalens, Arnaud Platz)

AVIO, RRD, RR, SNM, VOLVO
(+ MTU, TM ???)

NMS: Gleiwitz (PL), Budapest (H)

Cooling fundamentals including
radiation, liner materials, CMC,
near-wall effects, multi-physics, ...
Spray break-up, fuel types, tbd ...

Budget: ~ 9 MEUR (estimate)
EC funding: ~ 70 %
2006 – 2010 (48 months)





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ELECT is wishing you success

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